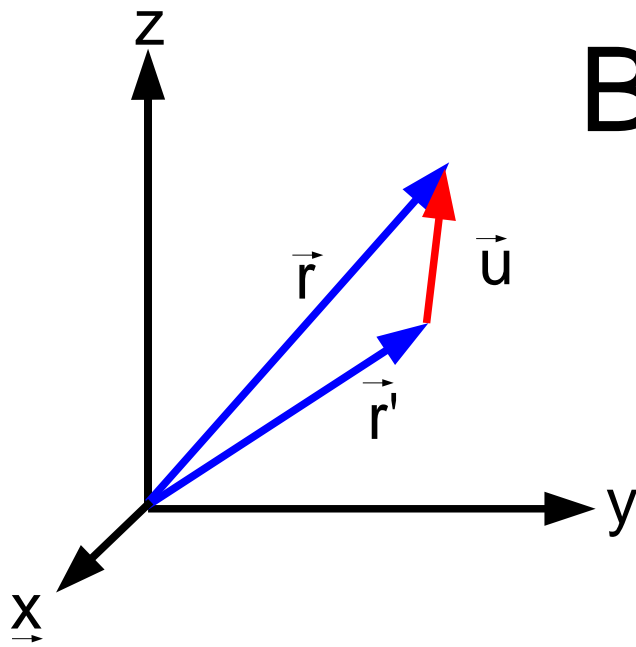


Formulation of the  
displacement-based  
Finite Element Method and  
General Convergence Results

# Basics of Elasticity Theory



$\vec{r}$  : before deformation  
 $\vec{r}'$  : after deformation  
 $\vec{u}$  : displacement

strain  $\varepsilon$ : measure of relative distortions

$$\varepsilon^T = (\varepsilon_{xx} \quad \varepsilon_{yy} \quad \varepsilon_{zz} \quad \gamma_{xy} \quad \gamma_{yz} \quad \gamma_{zx})$$

for small displacements  $\vec{u} = \begin{pmatrix} u \\ v \\ w \end{pmatrix}$  :

$$\varepsilon_{xx} = \frac{\partial u}{\partial x}, \quad \varepsilon_{yy} = \frac{\partial v}{\partial y}, \quad \varepsilon_{zz} = \frac{\partial w}{\partial z}$$

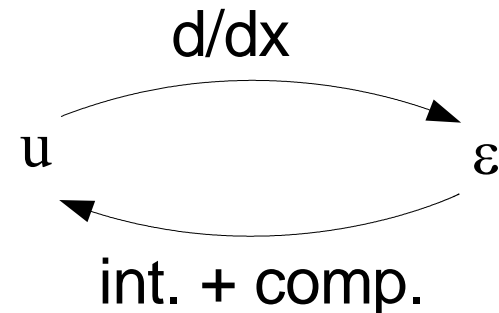
$$\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}, \quad \gamma_{yz} = \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y}$$

$$\gamma_{zx} = \frac{\partial w}{\partial x} + \frac{\partial u}{\partial z}$$

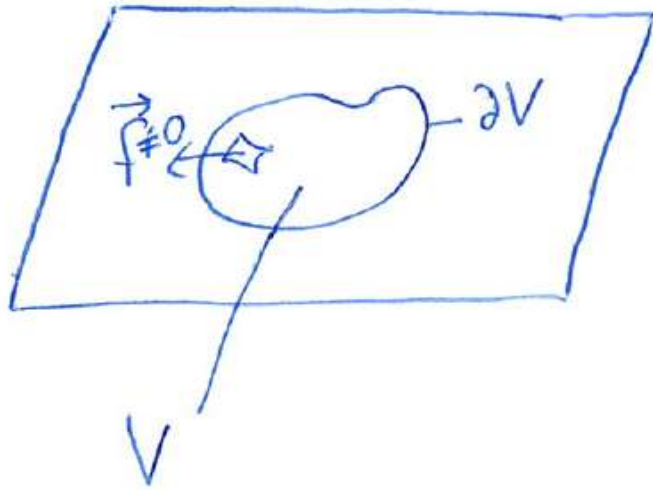
compatibility of the strain field:

$$\frac{\partial^2 \varepsilon_{ii}}{\partial x_k^2} + \frac{\partial^2 \varepsilon_{kk}}{\partial x_i^2} - 2 \frac{\partial^2 \varepsilon_{ik}}{\partial x_i \partial x_k} = 0$$

$$\frac{\partial^2 \varepsilon_{ik}}{\partial x_k \partial x_i} + \frac{\partial^2 \varepsilon_{kl}}{\partial x_i \partial x_k} - \frac{\partial^2 \varepsilon_{ii}}{\partial x_k^2} - \frac{\partial^2 \varepsilon_{kk}}{\partial x_i \partial x_i} = 0$$



# Basics of Elasticity Theory



deformed solid

deformation creates internal forces

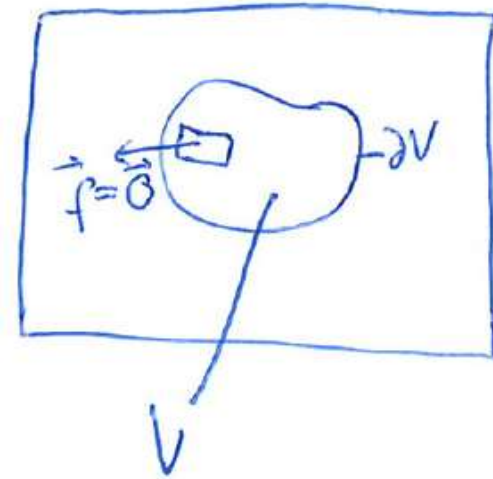
$$\tau = C \epsilon + \tau^I$$

$$\tau^I = [\tau_{xx} \quad \tau_{yy} \quad \tau_{zz} \quad \tau_{xy} \quad \tau_{yz} \quad \tau_{zx}]$$

C: material matrix (symmetric)

$\epsilon$ : strain

$\tau^I$ : stress b.c.



solid in equilibrium

$$\nabla \cdot \tau = 0$$

$$\tau = C \epsilon + \tau^I$$

$$\nabla \cdot \tau + f = 0$$

the problem of elasticity:

$$G \cdot \left\{ \nabla \cdot \nabla \vec{u} - \frac{1}{1-2\nu} \nabla (\nabla \cdot \vec{u}) \right\} + \rho \vec{f} = 0$$

find  $u$  in the presence of boundary conditions:

$$u|_{S_u} \quad (\text{essential})$$

$$f|_{S_f} \quad (\text{natural})$$

$$\epsilon_{xx} = \frac{\partial u}{\partial x}, \quad \epsilon_{yy} = \frac{\partial v}{\partial y}, \quad \epsilon_{zz} = \frac{\partial w}{\partial z}$$

$$\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}, \quad \gamma_{yz} = \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y}$$

$$\gamma_{zx} = \frac{\partial w}{\partial x} + \frac{\partial u}{\partial z}$$

# The principle of virtual displacements

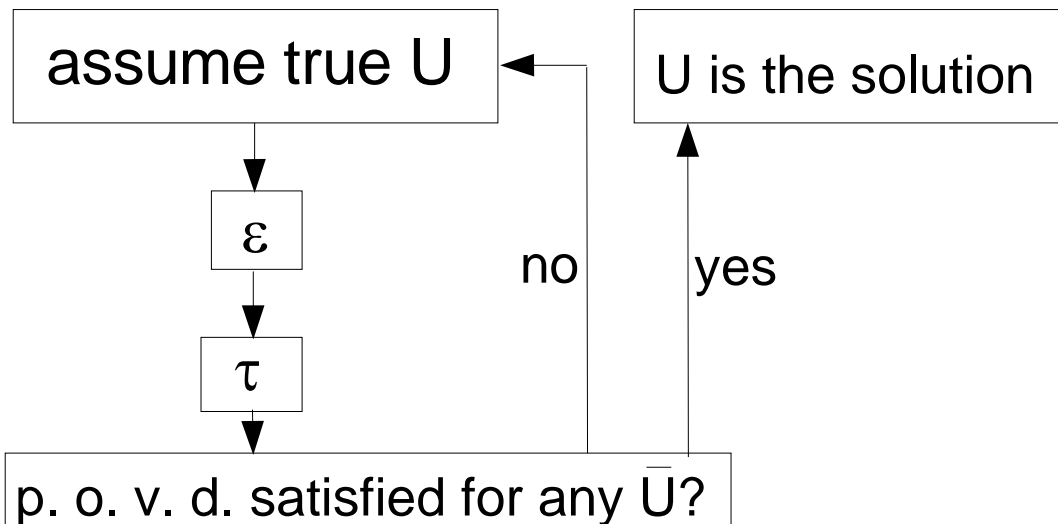
for any  $\bar{U}$  that obey the b.c. and the true equilibrium stress field:

internal virtual work

virtual work by external forces

$$\int_V \bar{\boldsymbol{\varepsilon}}^T \boldsymbol{\tau} \, dV = \int_V \bar{\mathbf{U}}^T \mathbf{f}^B \, dV + \int_{S_f} \bar{\mathbf{U}}^{S_f T} \mathbf{f}^{S_f} \, dS + \sum_i \bar{\mathbf{U}}^{i T} \mathbf{R}_C^i$$

stress field in equilibrium with  
virt. strain derived from virt. displacements  $\bar{U}$



implicitly contains:

- force boundary conditions
- direct stiffness method (in FEM formulation)

# FEM: domain discretization

global, nodal displacements:

$$\hat{\mathbf{U}}^T = [U_1 \ V_1 \ W_1 \ U_2 \ V_2 \ W_2 \ \dots \ U_N \ V_N \ W_N]$$

interpolation matrix  $\mathbf{H}^{(m)}$ :

$$\mathbf{u}^{(m)}(x, y, z) = \mathbf{H}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

strain-displacement matrix  $\mathbf{B}^{(m)}$ :

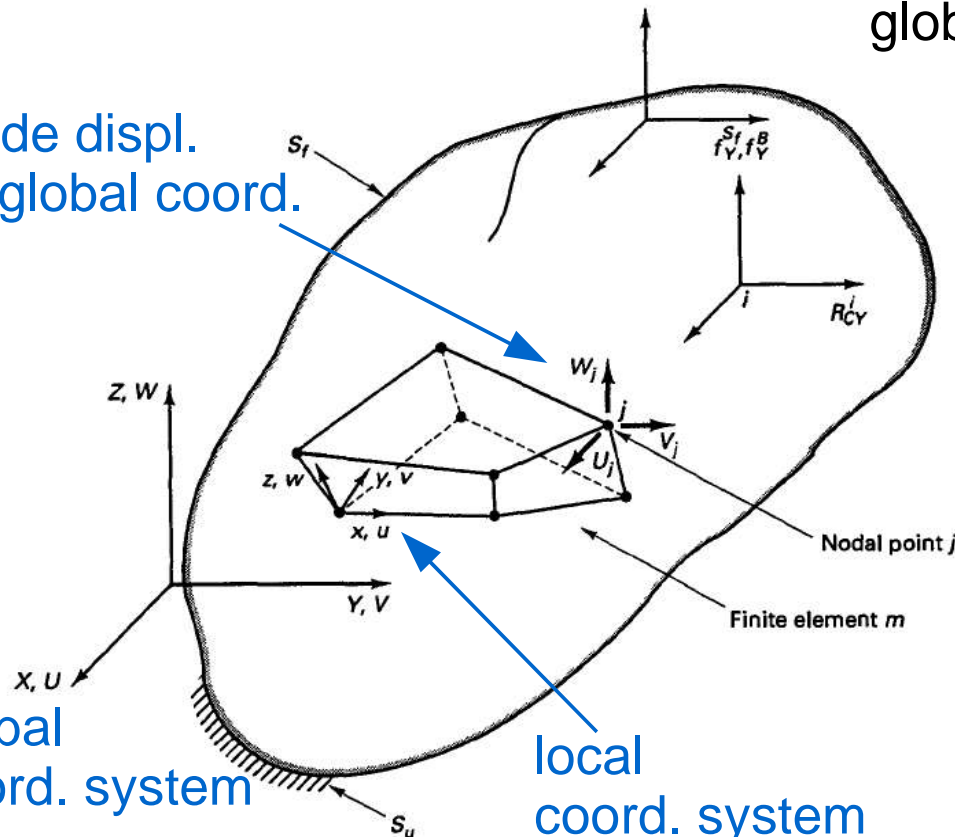
$$\boldsymbol{\epsilon}^{(m)}(x, y, z) = \mathbf{B}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

it holds:  $\boldsymbol{\tau}^{(m)} = \mathbf{C}^{(m)} \boldsymbol{\epsilon}^{(m)} + \boldsymbol{\tau}^I(m)$

principle of virtual displacements for the discretized body:

$$\begin{aligned} \sum_m \int_{V^{(m)}} \bar{\boldsymbol{\epsilon}}^{(m)T} \boldsymbol{\tau}^{(m)} dV^{(m)} &= \sum_m \int_{V^{(m)}} \bar{\mathbf{u}}^{(m)T} \mathbf{f}^{B(m)} dV^{(m)} \\ &+ \sum_m \int_{S_1^{(m)}, \dots, S_q^{(m)}} \bar{\mathbf{u}}^{S(m)T} \mathbf{f}^{S(m)} dS^{(m)} + \sum_i \bar{\mathbf{u}}^i T \mathbf{R}_C^i \end{aligned}$$

node displ. in global coord.



Nodal point  $j$

Finite element  $m$

local coord. system of  $m$ -th element

global coord. system

solid decomposed into assembly of finite elements:  
no overlap no holes  
nodal points coincide

# FEM: matrix equations

ansatz for displacement field

$$\boldsymbol{\epsilon}^{(m)}(x, y, z) = \mathbf{B}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

$$\mathbf{u}^{(m)}(x, y, z) = \mathbf{H}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

ansatz for virt. displacement field

$$\bar{\mathbf{u}}^{(m)}(x, y, z) = \mathbf{H}^{(m)}(x, y, z) \bar{\mathbf{U}}$$

$$\bar{\boldsymbol{\epsilon}}^{(m)}(x, y, z) = \mathbf{B}^{(m)}(x, y, z) \bar{\mathbf{U}}$$

discretized principle of virt. displ.

discrete displacement field to be determined

$$\begin{aligned} \bar{\mathbf{U}}^T \left[ \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)T} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} \right] \hat{\mathbf{U}} &= \bar{\mathbf{U}}^T \left[ \left\{ \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{B(m)} dV^{(m)} \right\} \right. \\ &+ \left. \left\{ \sum_m \int_{S_1^{(m)}, \dots, S_q^{(m)}} \mathbf{H}^{S(m)T} \mathbf{f}^{S(m)} dS^{(m)} \right\} \right. \\ &- \left. \left\{ \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)T} \boldsymbol{\tau}^{I(m)} dV^{(m)} \right\} + \mathbf{R}_C \right] \end{aligned}$$

$\hat{\mathbf{U}}$  solution when p.o.v.d. fulfilled  $\forall \bar{\mathbf{U}}$  especially:  $\bar{\mathbf{U}} = \mathbf{e}_i$

with  $\mathbf{U} \equiv \hat{\mathbf{U}}$  one obtains:  $\mathbf{K}\mathbf{U} = \mathbf{R}$ , where  $\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S - \mathbf{R}_I + \mathbf{R}_C$

$$\mathbf{K} = \sum_m \underbrace{\int_{V^{(m)}} \mathbf{B}^{(m)T} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)}}_{= \mathbf{K}^{(m)}}$$

implicit direct stiffness

$$\mathbf{R}_B = \sum_m \underbrace{\int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{B(m)} dV^{(m)}}_{= \mathbf{R}_B^{(m)}} \quad \mathbf{R}_S = \sum_m \underbrace{\int_{S_1^{(m)}, \dots, S_q^{(m)}} \mathbf{H}^{S(m)T} \mathbf{f}^{S(m)} dS^{(m)}}_{= \mathbf{R}_S^{(m)}}$$

boundary forces contained

$$\mathbf{R}_I = \sum_m \underbrace{\int_{V^{(m)}} \mathbf{B}^{(m)T} \boldsymbol{\tau}^{I(m)} dV^{(m)}}_{= \mathbf{R}_I^{(m)}}$$

# FEM: matrix equations

dynamic case with dissipative forces  $\sim dU/dt$ :

modified body forces:  $\mathbf{R}_B = \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)T} [\mathbf{f}^{B(m)} - \rho^{(m)} \mathbf{H}^{(m)} \ddot{\mathbf{U}} - \kappa^{(m)} \mathbf{H}^{(m)} \dot{\mathbf{U}}] dV^{(m)}$


  
 inertial forces    damping forces

define:

$$\mathbf{C} = \sum_m \underbrace{\int_{V^{(m)}} \kappa^{(m)} \mathbf{H}^{(m)T} \mathbf{H}^{(m)} dV^{(m)}}_{= \mathbf{C}^{(m)}} \quad \mathbf{M} = \sum_m \underbrace{\int_{V^{(m)}} \rho^{(m)} \mathbf{H}^{(m)T} \mathbf{H}^{(m)} dV^{(m)}}_{= \mathbf{M}^{(m)}}$$

**→  $\mathbf{M} \ddot{\mathbf{U}} + \mathbf{C} \dot{\mathbf{U}} + \mathbf{K} \mathbf{U} = \mathbf{R}$**

$$\boldsymbol{\epsilon}^{(m)}(x, y, z) = \mathbf{B}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

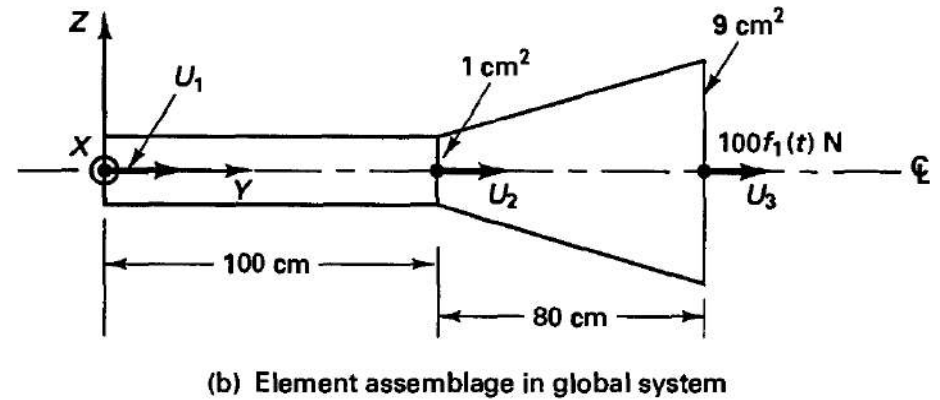
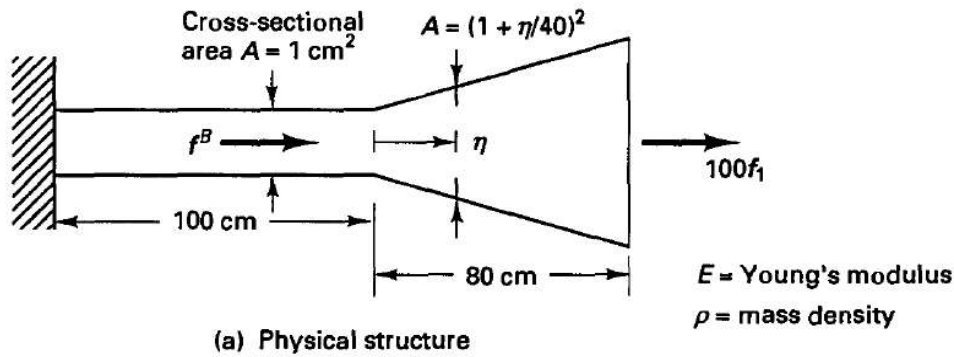
$$\mathbf{u}^{(m)}(x, y, z) = \mathbf{H}^{(m)}(x, y, z) \hat{\mathbf{U}}$$

$$\bar{\mathbf{u}}^{(m)}(x, y, z) = \mathbf{H}^{(m)}(x, y, z) \bar{\hat{\mathbf{U}}}$$

$$\bar{\boldsymbol{\epsilon}}^{(m)}(x, y, z) = \mathbf{B}^{(m)}(x, y, z) \bar{\hat{\mathbf{U}}}$$

—————▶ K and M are symmetric

# FEM: Two-element bar assemblage



$$u^{(m)}(x) = H^{(m)}(x) \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}$$

$$= \begin{bmatrix} H_1^{(m)}(x) & H_2^{(m)}(x) & H_3^{(m)}(x) \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}$$

two 2-point elements, polynomial ansatz

$$\rightarrow u^{(m)}(x) \text{ linear: } H_i^{(m)}(x) = \alpha_i^{(m)} + \beta_i^{(m)} x$$

interpolation condition:  $u^{(1)}(0) = u_1, \forall u_2$

$$H_3^{(1)}(x) = 0$$

$m = 1$

$$u^{(1)}(100) = u_2, \forall u_1$$

$$u^{(1)}(0) = \alpha_1^{(1)} u_1 + \alpha_2^{(1)} u_2 = u_1, \forall u_2$$

$$\Rightarrow \alpha_1^{(1)} = 1, \alpha_2^{(1)} = 0$$

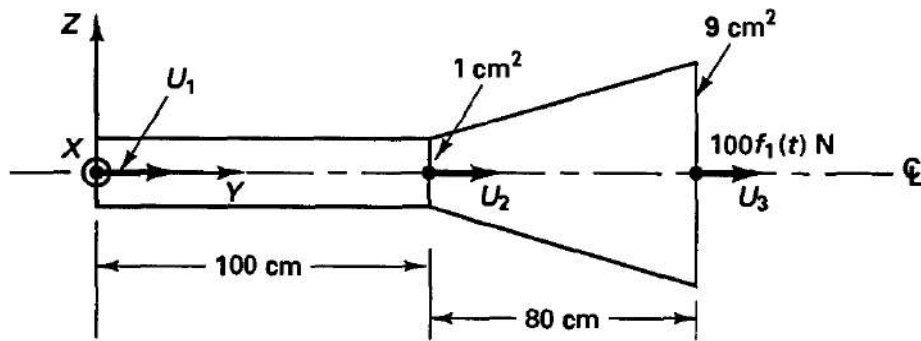
$$u^{(1)}(100) = (1 + 100 \cdot \beta_1^{(1)}) u_1 + \beta_2^{(1)} u_2 = u_2, \forall u_1$$

$$\Rightarrow \beta_2^{(1)} = 1, \beta_1^{(1)} = -\frac{1}{100}$$

$$\Rightarrow u^{(1)}(x) = \left(1 - \frac{x}{100}\right) u_1 + \frac{x}{100} u_2 \quad \Rightarrow H^{(1)}(x) = \begin{bmatrix} \left(1 - \frac{x}{100}\right) & \frac{x}{100} & 0 \end{bmatrix}$$



# FEM: Two-element bar assemblage



(b) Element assemblage in global system

calculate  $B^{(1)}(x)$  for given  $u^{(1)}(x)$ :

$$\begin{aligned} \epsilon^{(m)}(x) &= [B_1^{(m)}(x) \quad B_2^{(m)}(x) \quad B_3^{(m)}(x)] \begin{bmatrix} U_1 \\ U_2 \\ U_3 \end{bmatrix} \\ &= B_1^{(m)}(x)U_1 + B_2^{(m)}(x)U_2 + B_3^{(m)}(x)U_3 \end{aligned}$$

$$m = 1: \quad \epsilon^{(1)}(x) = \frac{d}{dx} u^{(1)}(x) = -\frac{1}{100} U_1 + \frac{1}{100} U_2$$

$$\Rightarrow B^{(1)}(x) = \left[ -\frac{1}{100} \quad \frac{1}{100} \quad 0 \right]$$

analogous:

$$\mathbf{H}^{(2)} = \begin{bmatrix} 0 & \left(1 - \frac{x}{80}\right) & \frac{x}{80} \end{bmatrix}$$

$$\mathbf{B}^{(2)} = \begin{bmatrix} 0 & -\frac{1}{80} & \frac{1}{80} \end{bmatrix}$$

material matrix:

$$\mathbf{C}^{(1)} = E; \quad \mathbf{C}^{(2)} = E$$

# FEM: Two-element bar assemblage

slow load application:  $\mathbf{KU} = \mathbf{R}$ ,  $\mathbf{R} = \mathbf{R}_B + \mathbf{R}_S - \mathbf{R}_I + \mathbf{R}_C$

$$\mathbf{K} = \sum_m \int_{V^{(m)}} \mathbf{B}^{(m)T} \mathbf{C}^{(m)} \mathbf{B}^{(m)} dV^{(m)} = \mathbf{K}^{(m)}$$

$$\mathbf{R}_B = \sum_m \int_{V^{(m)}} \mathbf{H}^{(m)T} \mathbf{f}^{(m)} dV^{(m)}$$

$$\mathbf{K} = \int_0^{100} \mathbf{B}^{(1)T} \mathbf{C}^{(1)} \mathbf{B}^{(1)} dx + \int_0^{80} \mathbf{B}^{(2)T} \mathbf{C}^{(2)} \mathbf{B}^{(2)} dx$$

$$= C^{(1)} \int_0^{100} A_1 dx \begin{bmatrix} -1/100 \\ 1/100 \\ 0 \end{bmatrix} \begin{bmatrix} -1/100 & 1/100 & 0 \end{bmatrix} dx + C^{(2)} \int_0^{80} A_1 \left(1 + \frac{x}{40}\right)^2 dx \begin{bmatrix} 0 \\ -1/80 \\ 1/80 \end{bmatrix} \begin{bmatrix} 0 & -1/80 & 1/80 \end{bmatrix}$$

$$= C^{(1)} A_1 \cdot 100 \cdot \begin{bmatrix} 1/100^2 & -1/100^2 & 0 \\ -1/100^2 & 1/100^2 & 0 \\ 0 & 0 & 0 \end{bmatrix} + C^{(2)} A_1 \int_0^{80} \left(1 + \frac{x}{40}\right)^2 dx \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1/80^2 & -1/80^2 \\ 0 & -1/80^2 & 1/80^2 \end{bmatrix}$$

$$= A_1 E \cdot \left( \frac{1}{100} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} + \frac{13}{240} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \right)$$

$$\mathbf{R}_B = \left\{ (1) \int_0^{100} \begin{bmatrix} 1 - \frac{x}{100} \\ \frac{x}{100} \\ 0 \end{bmatrix} (1) dx + \int_0^{80} \left(1 + \frac{x}{40}\right)^2 \begin{bmatrix} 1 - \frac{x}{80} \\ \frac{x}{80} \end{bmatrix} \left(\frac{1}{10}\right) dx \right\} f_2(t)$$

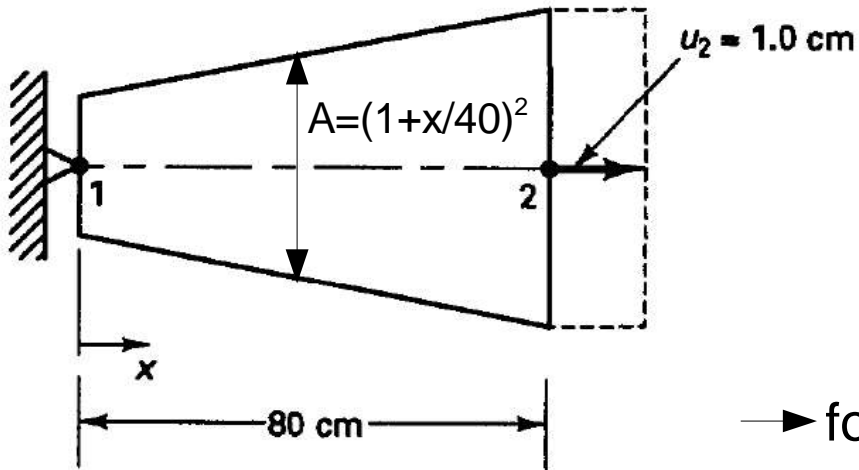
$$= \frac{1}{3} \begin{bmatrix} 150 \\ 186 \\ 68 \end{bmatrix} f_2(t)$$

$$\mathbf{R}_C = \begin{bmatrix} 0 \\ 0 \\ 100 \end{bmatrix} f_1(t)$$

$$\mathbf{R}_S = \mathbf{R}_I = 0$$

obtain  $U(t^*)$  by solving:  $\mathbf{KU}|_{t=t^*} = \mathbf{R}_B|_{t=t^*} + \mathbf{R}_C|_{t=t^*}$

# FEM: Exact stiffness vs. FEM approx.



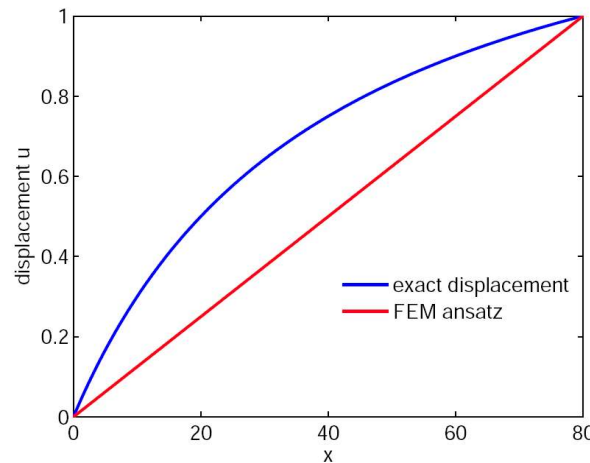
from variational principle:  $E \frac{d}{dx} \left( A \frac{du}{dx} \right) = 0$   
 exact solution w. b.c.  $u(0) = 0$  and  $u(80) = 1$ :

$$u = \frac{3}{2} \left( 1 - \frac{1}{1 + x/40} \right)$$

→ forces at the ends of the bar:

$$k_{12} = -EA \frac{du}{dx} \Big|_{x=0}$$

$$k_{22} = EA \frac{du}{dx} \Big|_{x=L}$$



$$\mathbf{K} = \frac{3}{80} E \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

(exact stiffness matrix)

$$\mathbf{K}_{\text{FEM}} = \frac{13E}{240} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix}$$

(FEM approx., prev. example)



displacement ansatz too rigid → overestimate of stiffness

# FEM: Displacement constraints

decompose  $U = \begin{bmatrix} U_a \\ U_b \end{bmatrix}$

$U_a$  : unconstrained

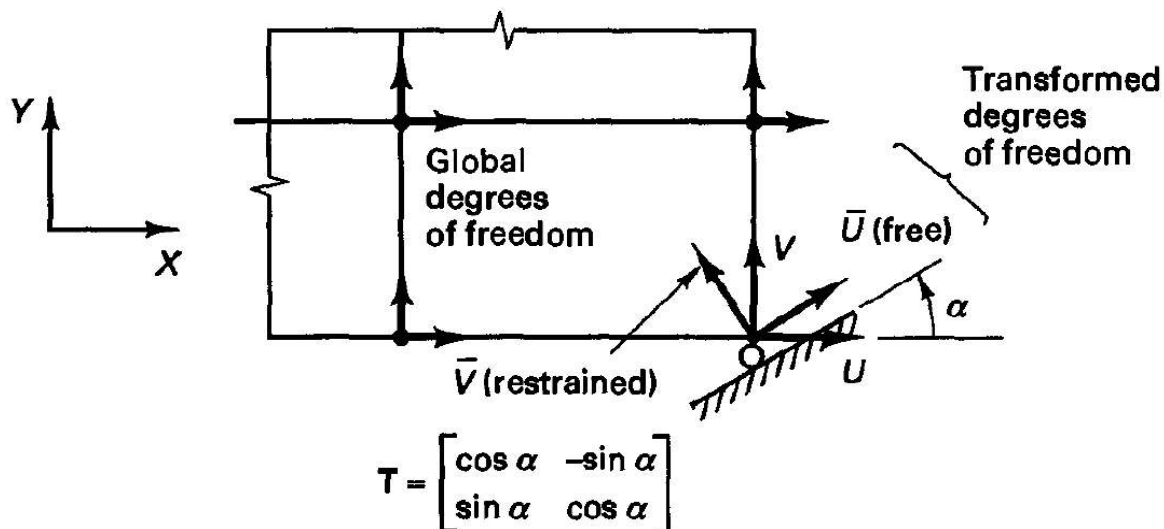
$U_b$  : constrained

FEM equations:

$$\begin{bmatrix} \mathbf{M}_{aa} & \mathbf{M}_{ab} \\ \mathbf{M}_{ba} & \mathbf{M}_{bb} \end{bmatrix} \begin{bmatrix} \ddot{U}_a \\ \ddot{U}_b \end{bmatrix} + \begin{bmatrix} \mathbf{K}_{aa} & \mathbf{K}_{ab} \\ \mathbf{K}_{ba} & \mathbf{K}_{bb} \end{bmatrix} \begin{bmatrix} U_a \\ U_b \end{bmatrix} = \begin{bmatrix} \mathbf{R}_a \\ \mathbf{R}_b \end{bmatrix}$$

$$\mathbf{M}_{aa}\ddot{U}_a + \mathbf{K}_{aa}U_a = \underbrace{\mathbf{R}_a - \mathbf{K}_{ab}U_b - \mathbf{M}_{ab}\ddot{U}_b}_{\text{modified force vector}}$$

if constraints don't coincide with  $U_b$ , try:  $U = \mathbf{T}\bar{U}$  where  $\bar{U}_b$  can be constrained



need to transform FEM matrices accordingly:

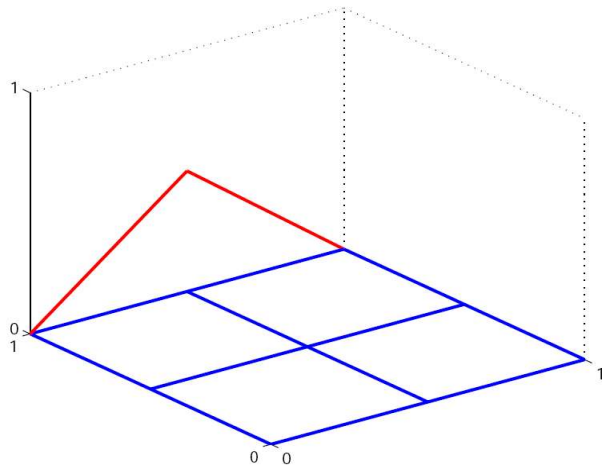
$$\bar{\mathbf{M}} = \mathbf{T}^T \mathbf{M} \mathbf{T}$$

$$\bar{\mathbf{K}} = \mathbf{T}^T \mathbf{K} \mathbf{T}$$

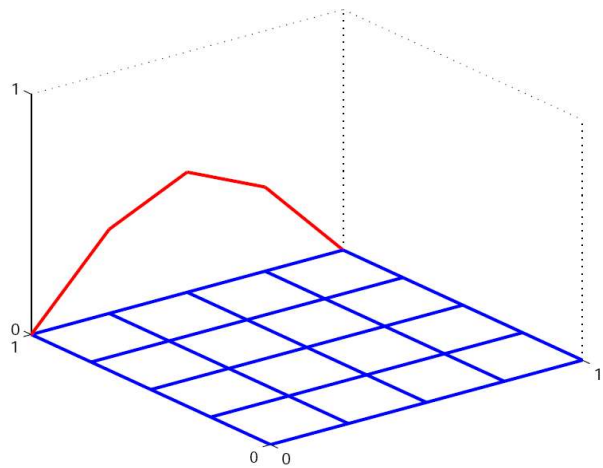
$$\bar{\mathbf{R}} = \mathbf{T}^T \mathbf{R}$$

$$\bar{\mathbf{M}}\ddot{\bar{U}} + \bar{\mathbf{K}}\bar{U} = \bar{\mathbf{R}}$$

# Refining the FEM solution



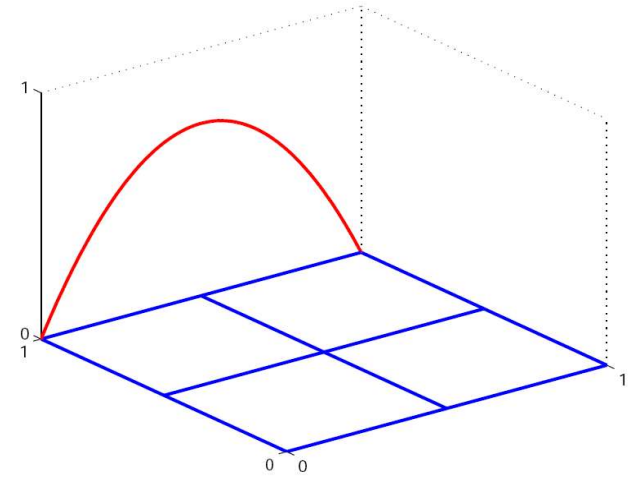
keep order  
of interpolation



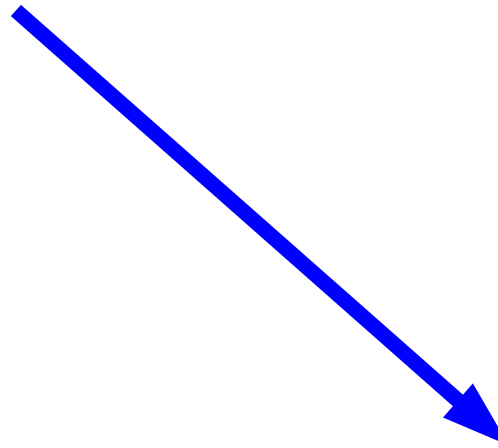
h refinement



keep element size



p refinement



h/p refinement:  
both simultaneously

# Convergence of a FEM solution

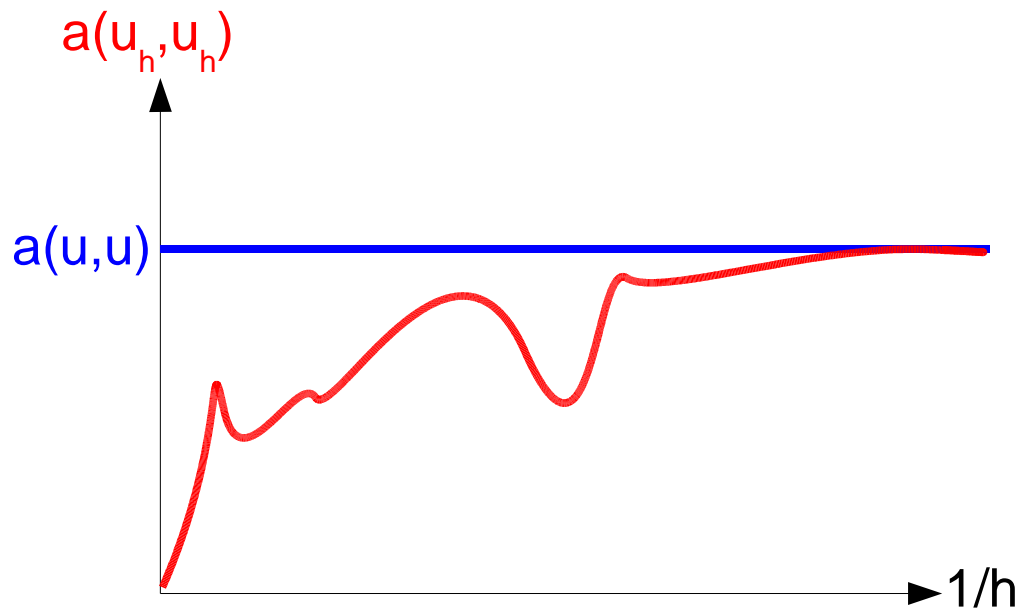
$$\int_V \bar{\boldsymbol{\epsilon}}^T \boldsymbol{\tau} dV = \int_{S_f} \bar{\mathbf{u}}^{S_f^T} \mathbf{f}^{S_f} dS + \int_V \bar{\mathbf{u}}^T \mathbf{f}^B dV \iff a(\mathbf{u}, \mathbf{v}) = (\mathbf{f}, \mathbf{v})$$

strain energy  $a(\mathbf{u}, \mathbf{v})$  pos. def. bilinear form

Convergence in the norm induced by a:

$$a(\mathbf{u}_h, \mathbf{u}_h) \rightarrow a(\mathbf{u}, \mathbf{u}) \quad \text{as } h \rightarrow 0 \quad \text{measures discretization and interpolation errors, only}$$

monotonic convergence:



necessary:

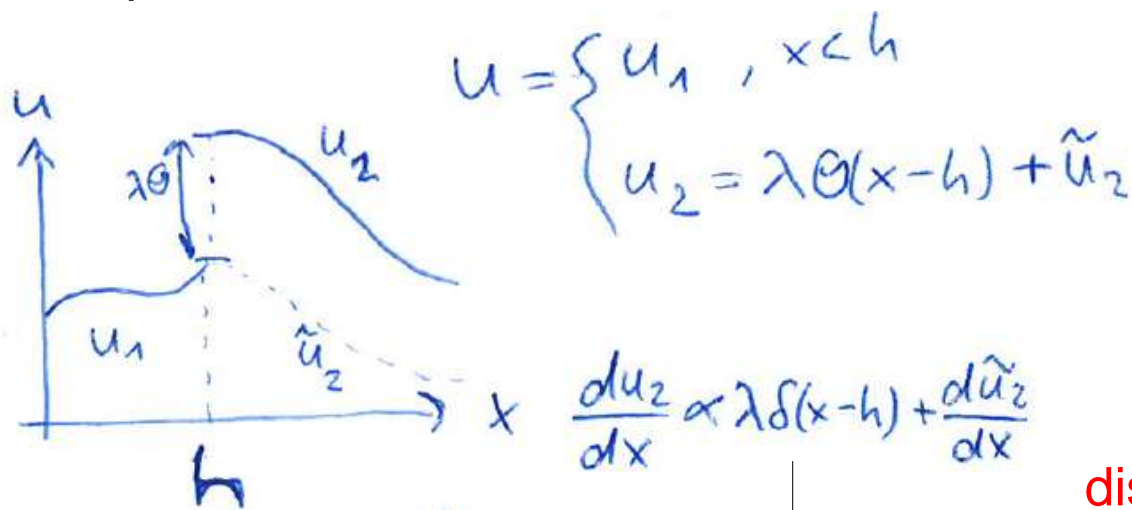
- compatibility of mesh and elements
- completeness of elements

# mon. Conv.: Compatibility

Compatibility:

continuity of  $u$  within and across element boundaries

Example:



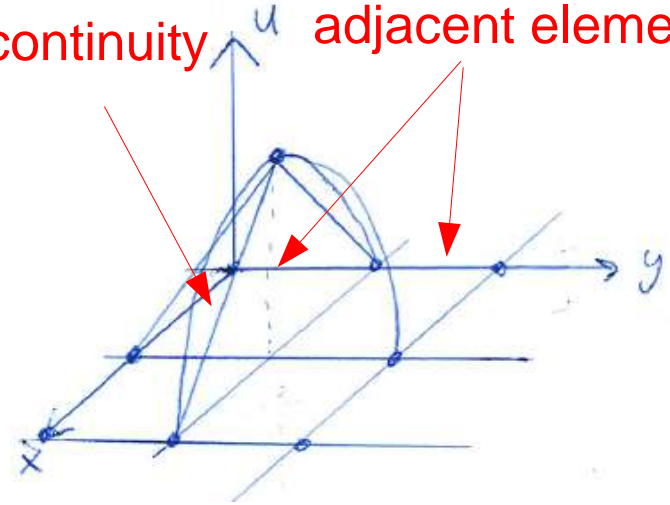
calculation of  $a(u_h, u_h)$ :

$$u \xrightarrow{d/dx} \varepsilon$$

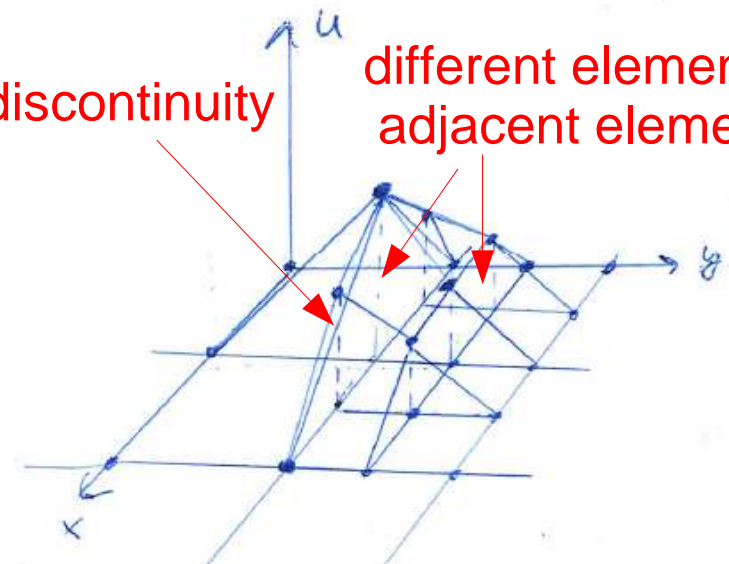
$$\varepsilon \xrightarrow{C} \tau$$

delta function gives unphysical contributions to  $a(u_h, u_h)$  and destroys monotonic convergence

discontinuity different interpolation is adjacent elements



discontinuity different element size in adjacent elements



# mon. Conv.: Completeness

Completeness:

element must be able to represent all rigid-body modes and states of constant strain

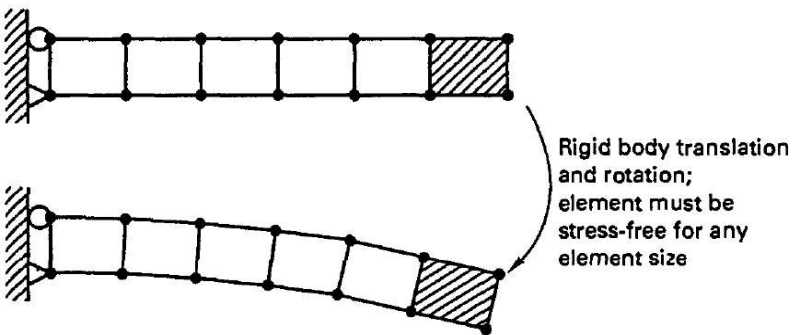
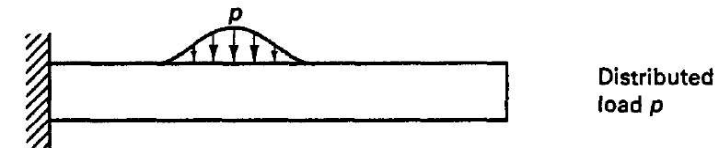
Number of rigid-body modes:

$$\mathbf{K}\boldsymbol{\phi} = \lambda\boldsymbol{\phi}$$

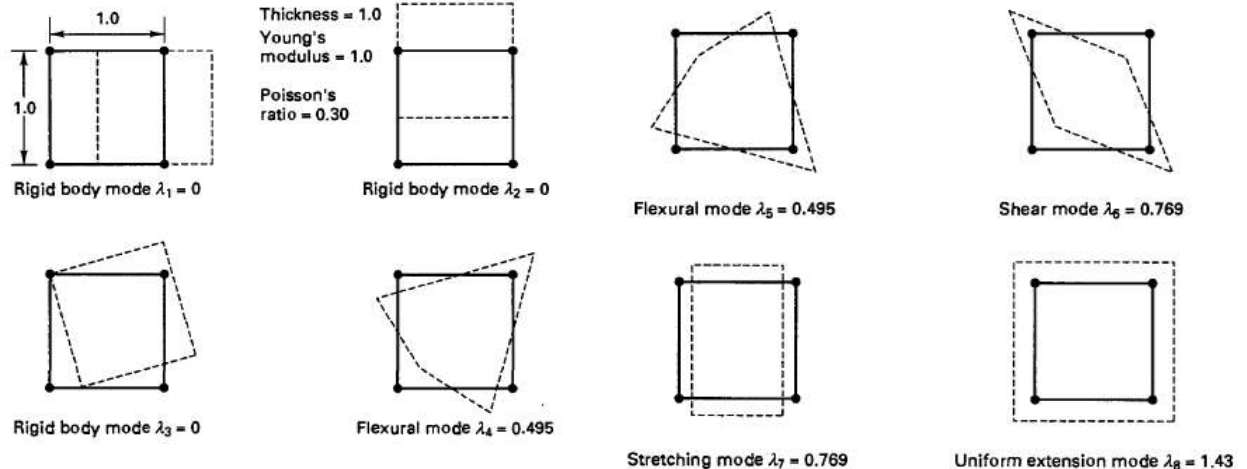
$$\longrightarrow \lambda_1, \dots, \lambda_n$$

$$\#rbm = \dim(\text{Ker}(\mathbf{K}))$$

Reason:



rbm determined by the eigenvectors of  $\mathbf{K}$ :

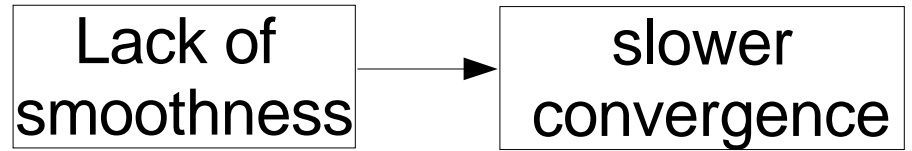
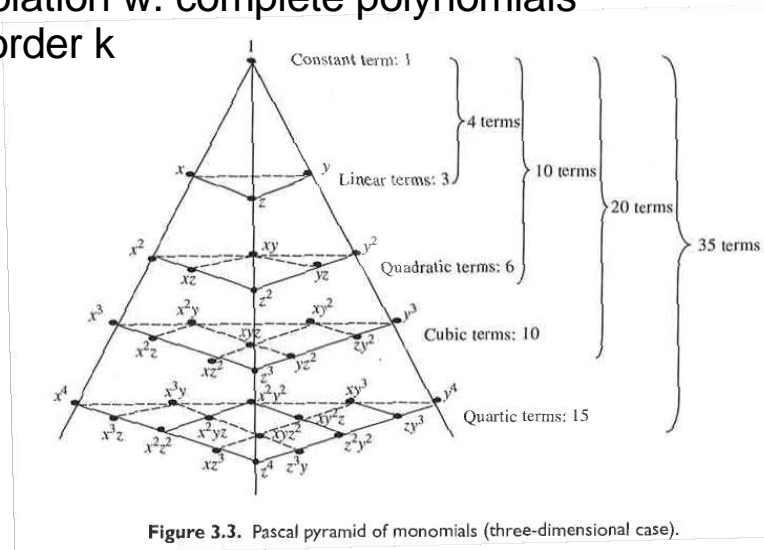




# Convergence rates

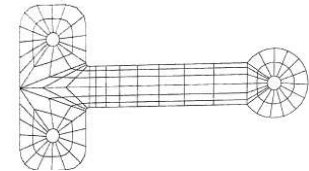
## Assumptions:

- interpolation w. complete polynomials up to order k



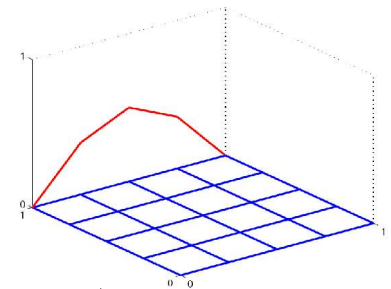
non-uniform grid:

$$\| \mathbf{u} - \mathbf{u}_h \|_1^2 \leq c \sum_m h_m^{2k} \| \mathbf{u} \|_{k+1,m}^2$$



h refinement:

$$\| \mathbf{u} - \mathbf{u}_h \|_1 \leq \frac{c}{(N)^{k/d}}$$



- exact solution is smooth enough, such that Sobolev-Norm of order k+1 is finite:

$$\| \mathbf{u} \|_{k+1} = \left\{ \int_{\text{Vol}} \left[ \sum_{i=1}^3 (u_i)^2 + \sum_{i=1}^3 \sum_{j=1}^3 \left( \frac{\partial u_i}{\partial x_j} \right)^2 + \sum_{n=2}^{k+1} \sum_{i=1}^3 \sum_{r+s+t=n} \left( \frac{\partial^n u_i}{\partial x_1^r \partial x_2^s \partial x_3^t} \right)^2 \right] d\text{Vol} \right\}^{1/2} < \infty$$

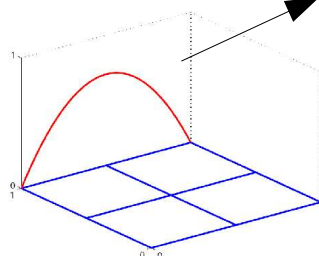
- uniform mesh

**h/p refinement:**

$$\| \mathbf{u} - \mathbf{u}_h \|_1 \leq \frac{c}{\exp [\beta(N)^r]}$$

Estimate (p-refinement):

$$\| \mathbf{u} - \mathbf{u}_h \|_0 \leq c h^{k+1} \| \mathbf{u} \|_{k+1}$$



Def.: (Sobolev-Norm of order 0)

$$(\| \mathbf{v} \|_0)^2 = \int_{\text{Vol}} \left( \sum_{i=1}^3 (v_i)^2 \right) d\text{Vol}$$

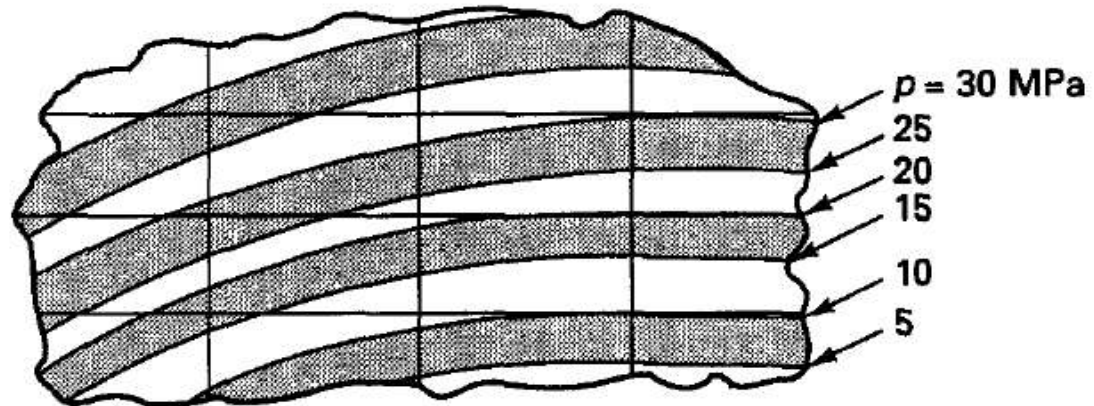
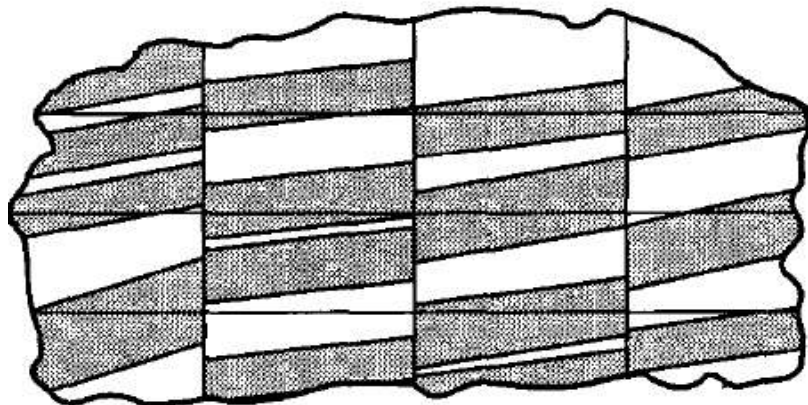
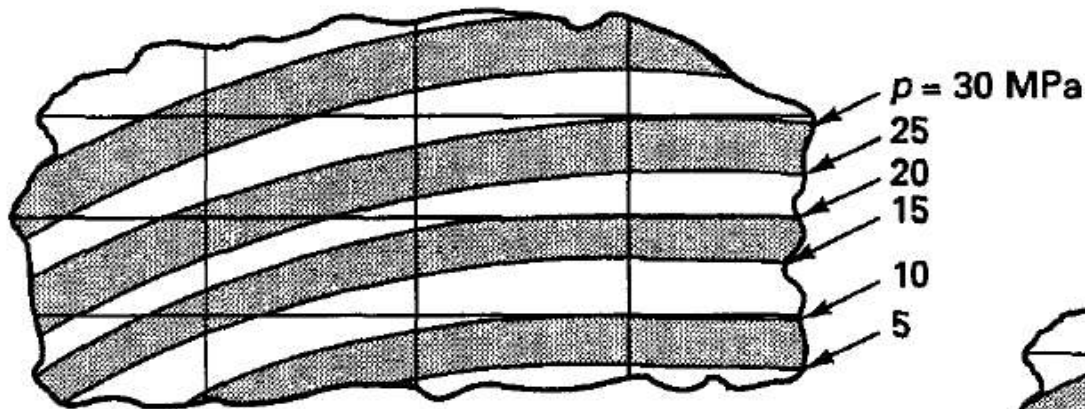
# Error assessment

FEM solution violates differential equilibrium:

$$\nabla \cdot \vec{\tau} + \vec{f} \neq 0$$

→ non-continuous stress possible

Graphical assessment:  
Isoband-plots



$\Delta p > 5 \text{ MPa}$ .

**Thank you for your attention!**